



# **Hardening Aircraft Against Terrorist Threats**

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# Participation in FAA Aircraft Hardening Program

(Completed)

- Development of Aircraft Structural Response Methodology
- Aircraft Response to Internal Explosive Detonation

(Ongoing)

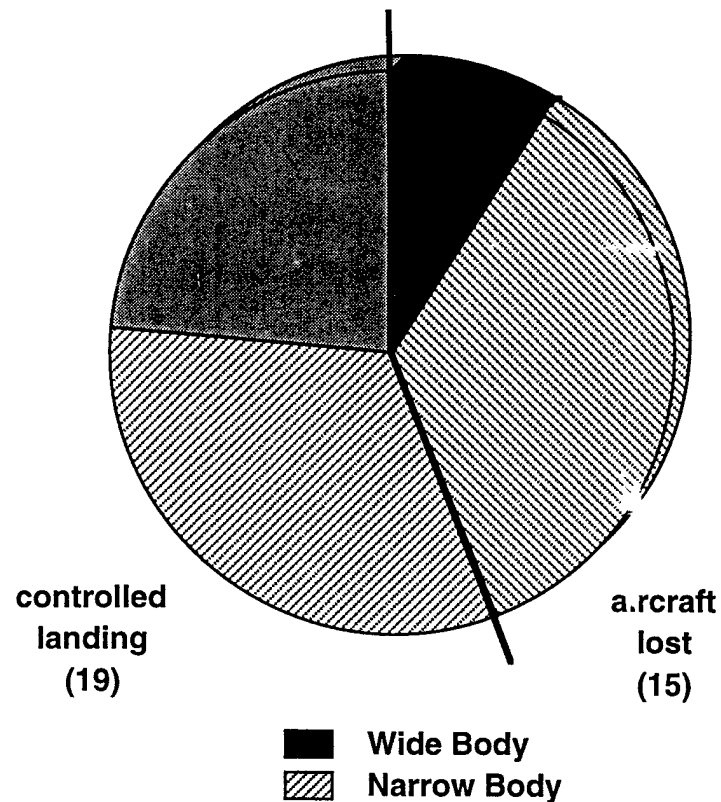
- Wide Body Aircraft Blast Test
- Reusable Blast Test Fixture



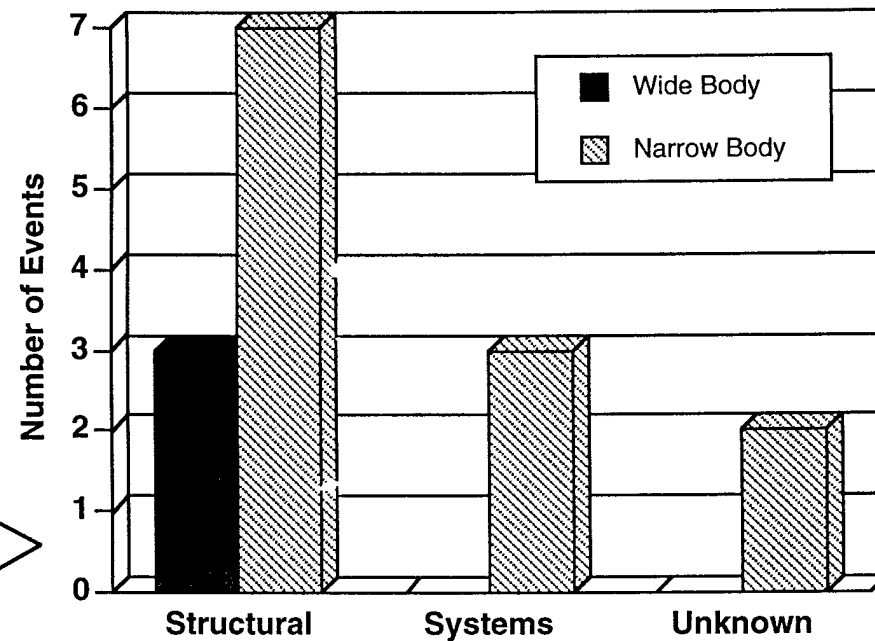
Aircraft	Bombing Attempts	Blast Events			% Survive Attempts	% Survive In-flight Blast
		Total	In flight	Catastrophic		
Total	81	58	34	15	81%	56%
Narrow-body	58	43	23	12	79%	48%
Wide-body	23	15	11	3	87%	73%
U.S. Events	10	4	1	0	100%	100%

# In-Flight Bomb Blast Events 1971 - 1995

*Event Outcome of 34  
In-Flight Incidents*



*Failure Modes of 15  
Aircraft Lost*

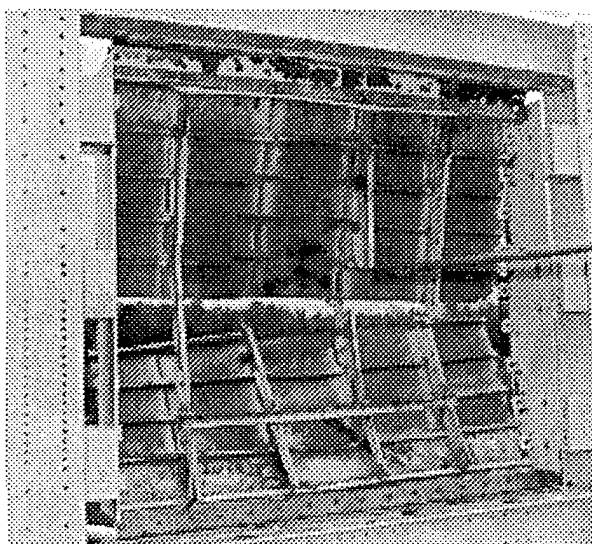
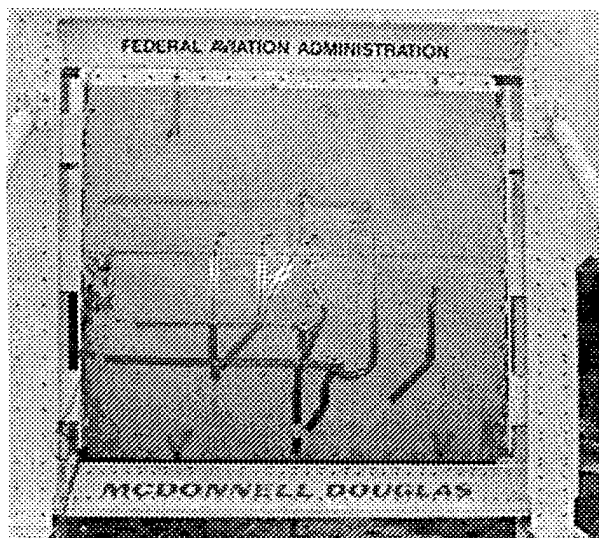


## Bombings Against Commercial Aviation

### Findings

- Bombings against commercial aviation are a worldwide problem.
  - Has already involved 33 countries and 40 airlines.
- Terrorists have knowledge of new methods to build bombs that may be more difficult to detect.
- 56% of aircraft survive in-flight bombings.
- Wide-body aircraft have a greater inherent tolerance to in flight bombings and 73% survive, compared to 48% of narrow-bodies.
  - Greater internal volume to absorb the blast
  - Greater structural surface area and multiple load paths to dissipate blast loads
  - Greater opportunity for separation of redundant systems

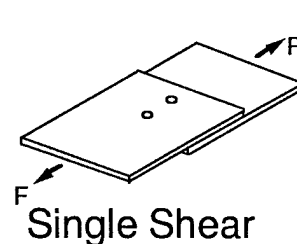
# Aircraft Hardening Program Materials Properties Database



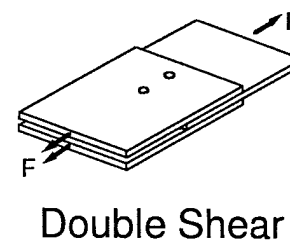
**PANEL TESTING**

- Mode I fracture toughness for 0.71 in. 2024-T3
- Rate-dependant constitutive relations (full range stress-strain diagrams)

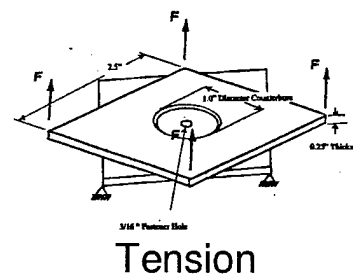
## **DYNAMIC MATERIALS PROPERTIES**



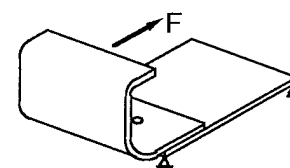
**Single Shear**



**Double Shear**

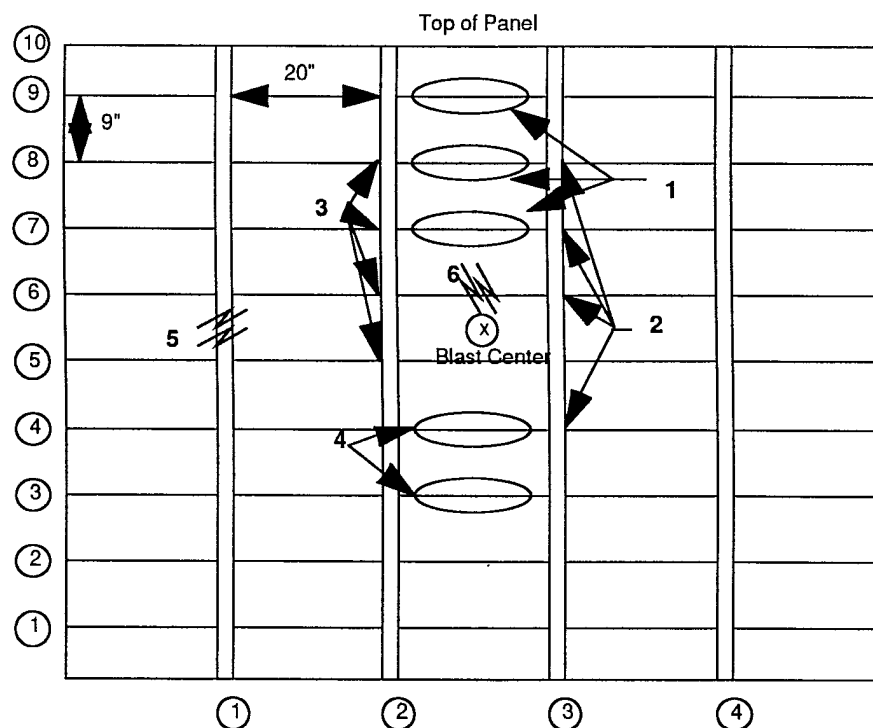





**Tension**



**Peeling load**

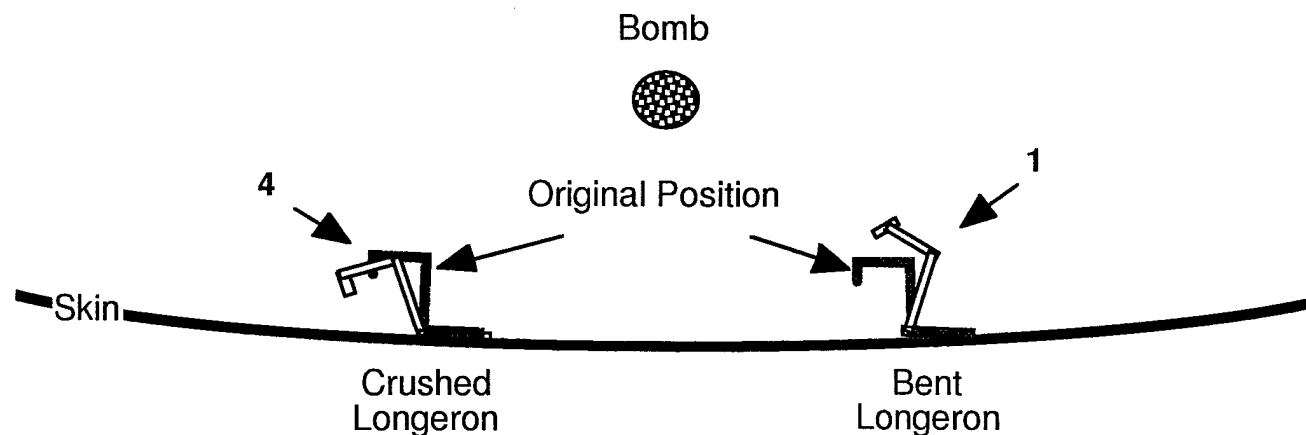
## **COMPONENT LEVEL TESTING**



Frames   
 Longerons   
 Broken structure 

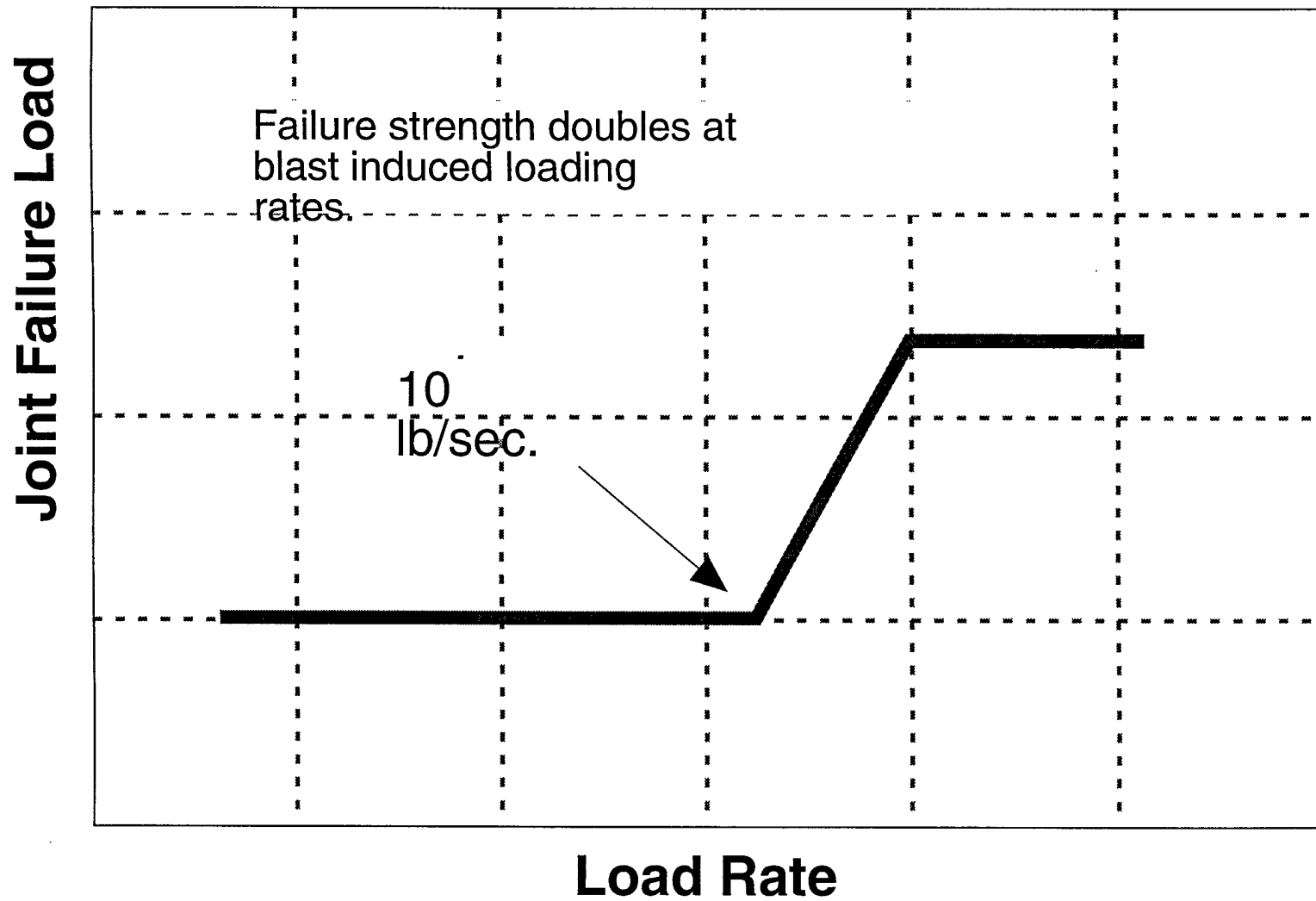
### Damage Description :

- 1 Areas where the longerons are bent upward, away from the blast wave.
- 2 Cleats between the frame and longeron broken.
- 3 Cleats between the frame and longeron broken.
- 4 Longerons slightly crushed from blast wave.
- 5 Frame 1 cracked in Mode III.
- 6 Longeron 6 also broke in Mode III, small area broken out.

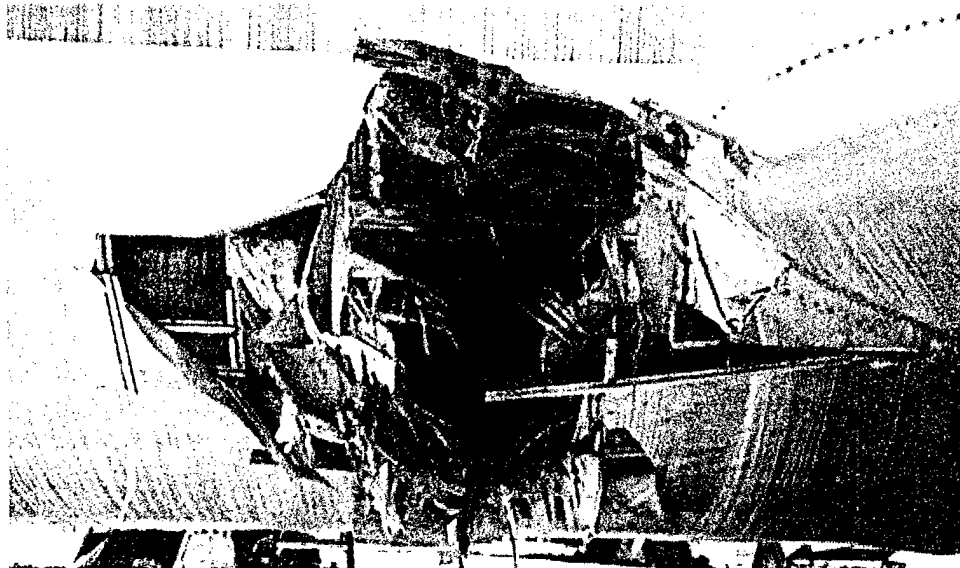




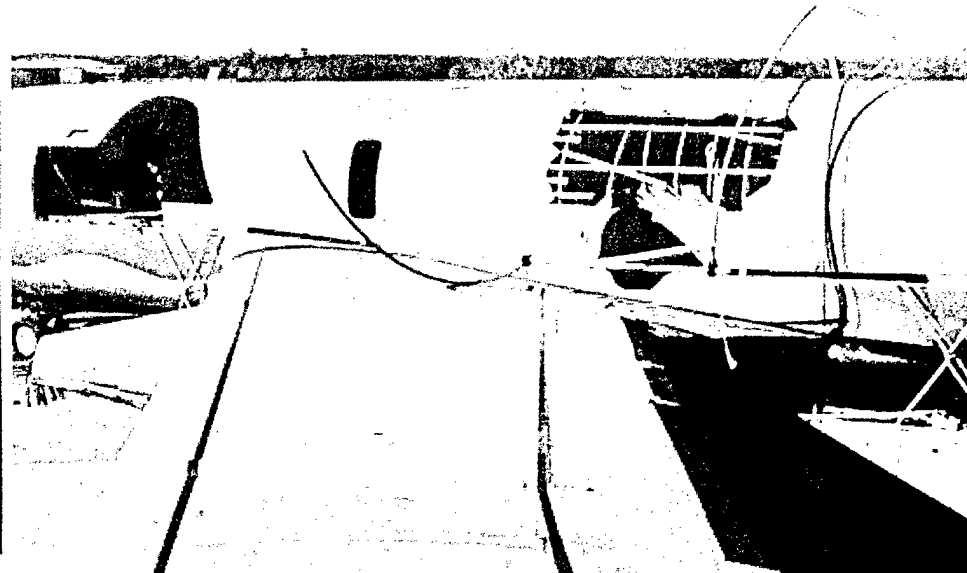
## Typical Test Results



# Aircraft Hardening Program Aircraft Test Database



B-707 Parametric Tests



KC-135 Pressurization Tests



C-880 Systems Vulnerability Tests



LD-3 Fragmentation Tests

# B-707 Narrow-Body Aircraft Tests

## Comparison of Damage Modes

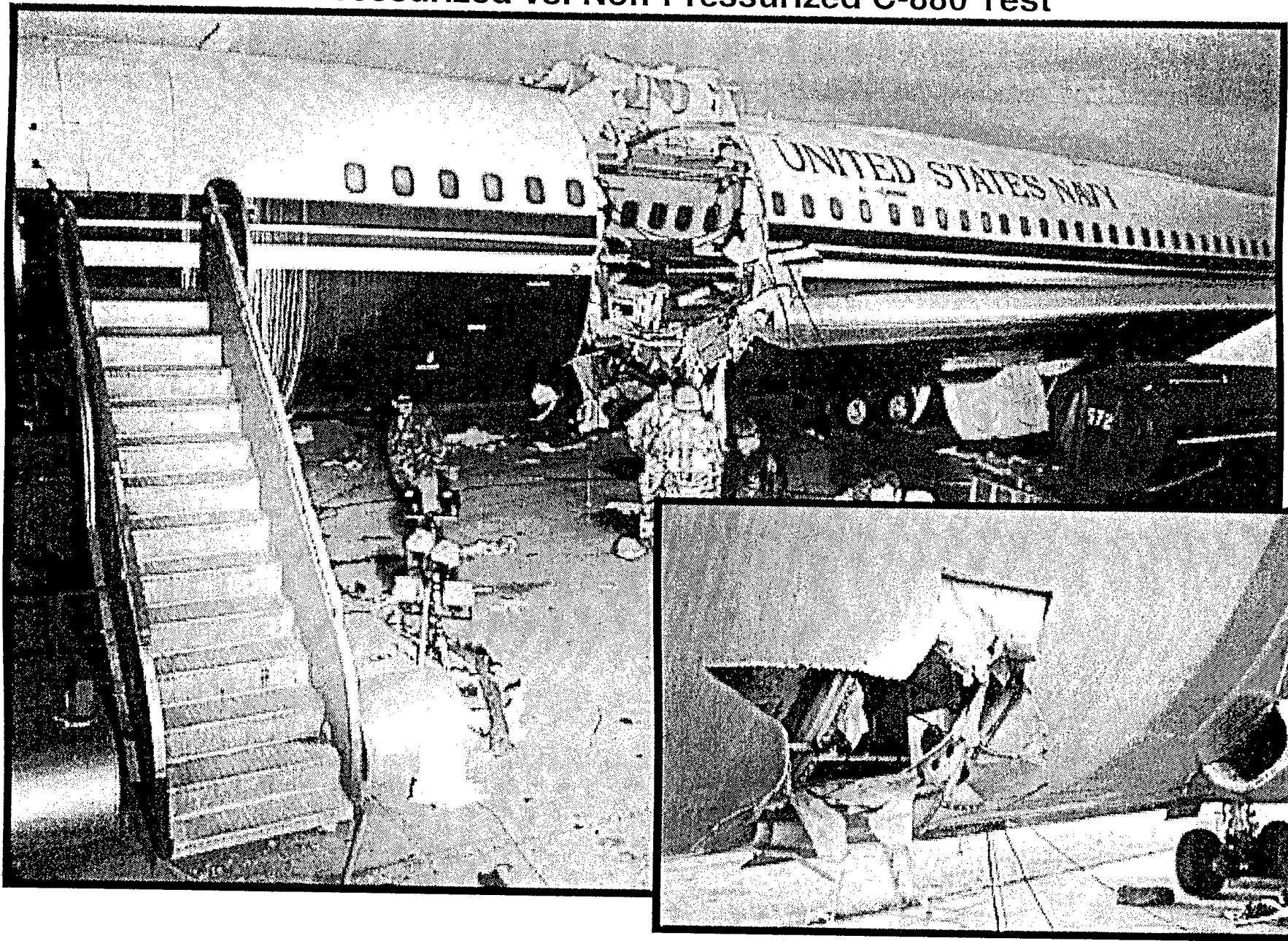


**Blast Damage from a Bare Charge**



**Blast Damage from a Fragmentation Charge  
(Single Suitcase)**

## Pressurized vs. Non-Pressurized C-880 Test

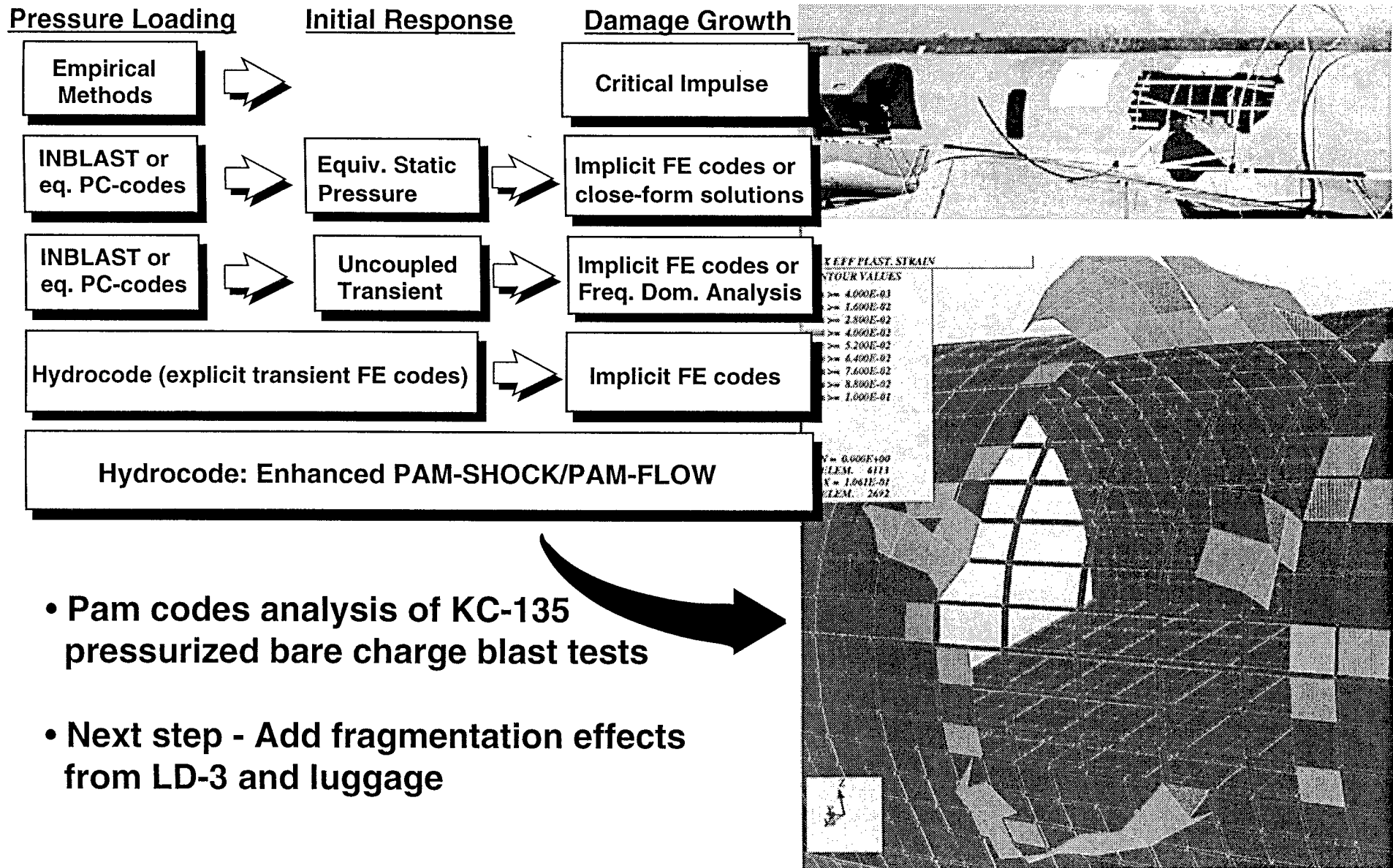


## LD-3 Threat Characterization Test Series





# Integrated Structural Response Methodologies



# Wide-Body Aircraft Blast Test



L-1011, Mobile Alabama

## PROGRAM OBJECTIVE:

To determine the minimum charge size and charge location to cause catastrophic damage, to a wide-body aircraft, from a bomb placed in a LD-3 luggage container.

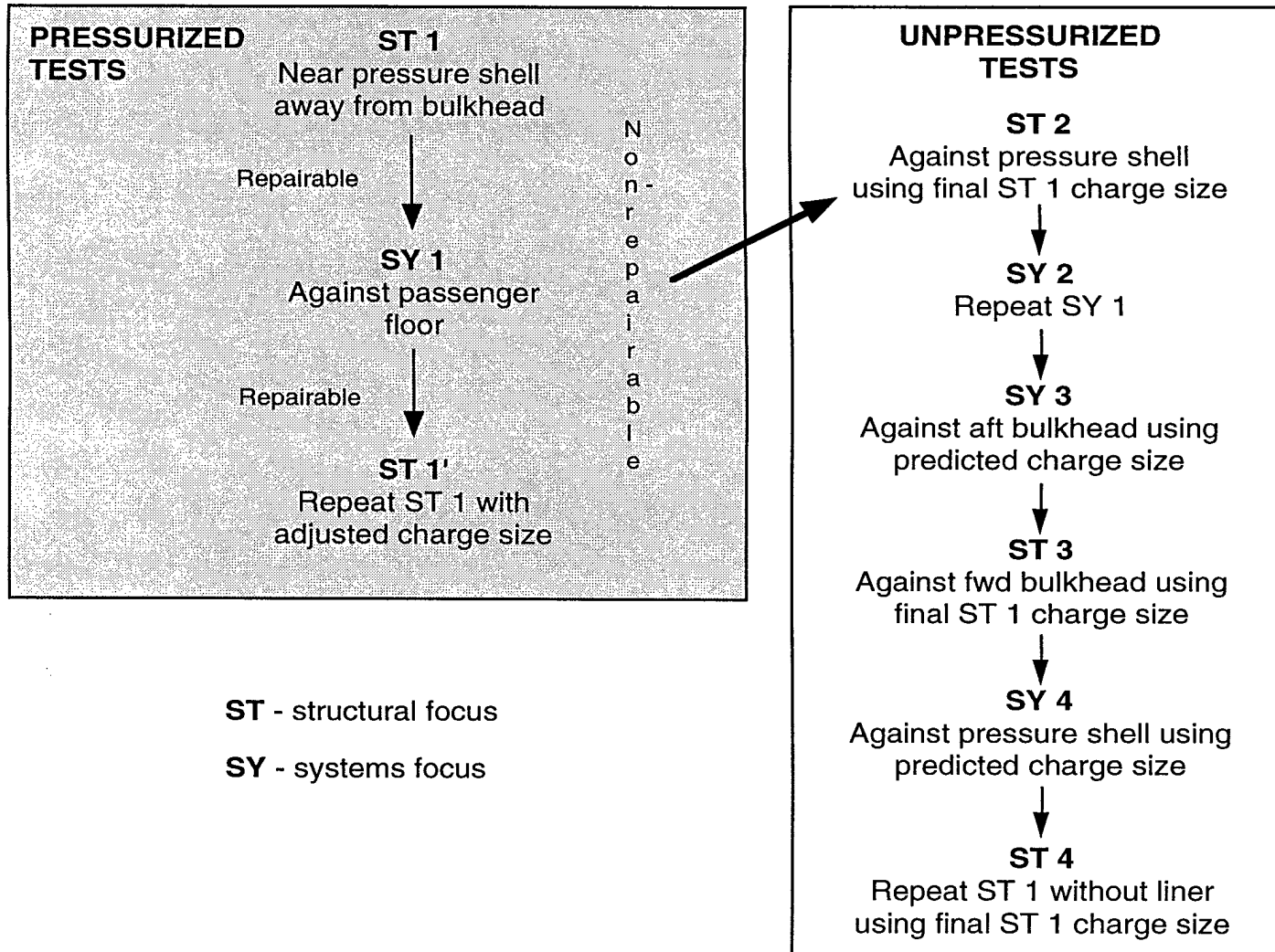
 **BOEING®**

## Test Simulates Operational Environment

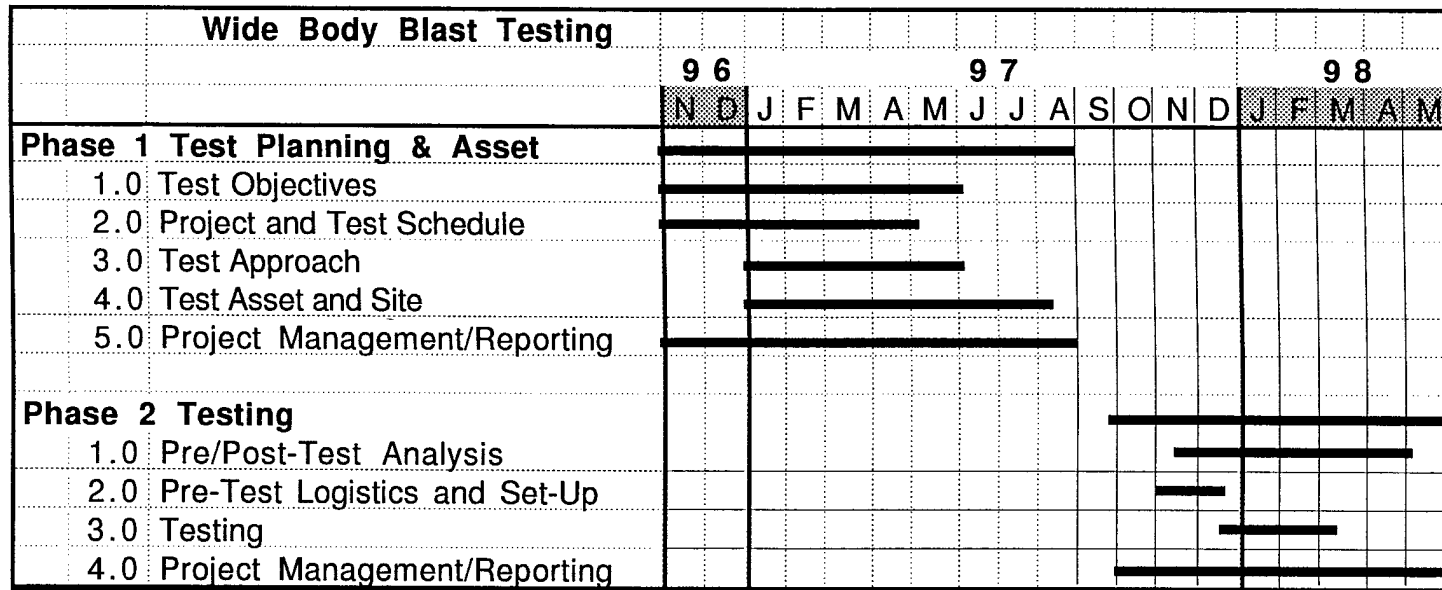
- Charge placed within luggage within LD-3 Container
- Container surrounded by other containers, all 75% full
- Representative delta pressures (~8.4 psi)



# Test Plan



# Program Schedule



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## Boeing Activities

Aircraft Preparation  
 Test Configuration  
 Instrumentation  
 General site support  
 Test Documentation and Analysis

## FAA Activities

Explosives Handling  
 Photo/Film

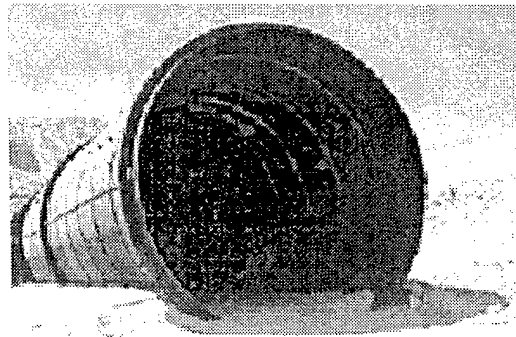


# Reusable Blast Test Fixture

## Objectives

- To provide an asset for gathering repeatable, realistic blast test data.
- To assess/verify hardening concepts; in particular, hardened containers.

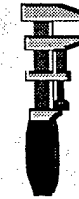
# Reusable Blast Test Fixture



EXISTING SHOCK TUBE



FIXTURE DESIGN



FABRICATION

**NMT**



EXPLOSIVE HANDLING  
AND TESTING

PROGRAM



REQUIREMENTS  
ANALYSIS

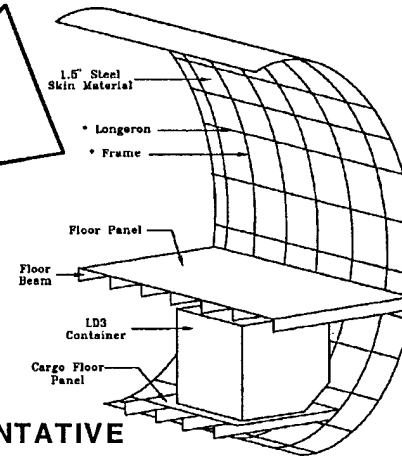


DESIGN ANALYSIS



TEST SECTION  
DESIGN

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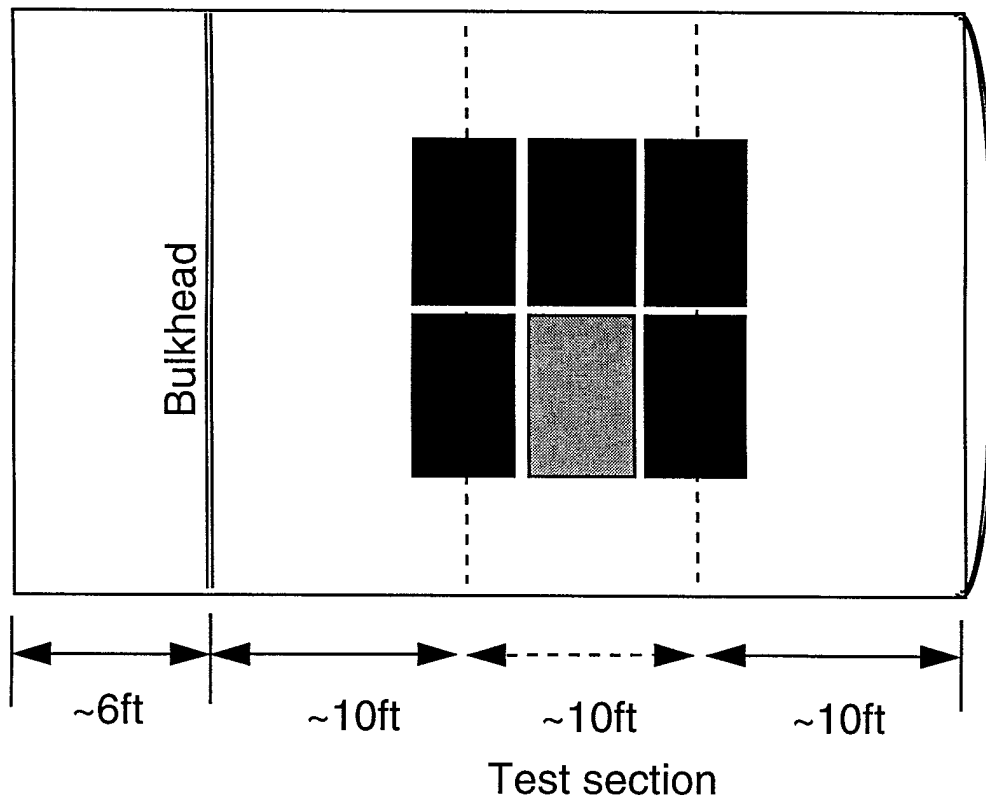


REPRESENTATIVE  
AIRCRAFT STRUCTURE

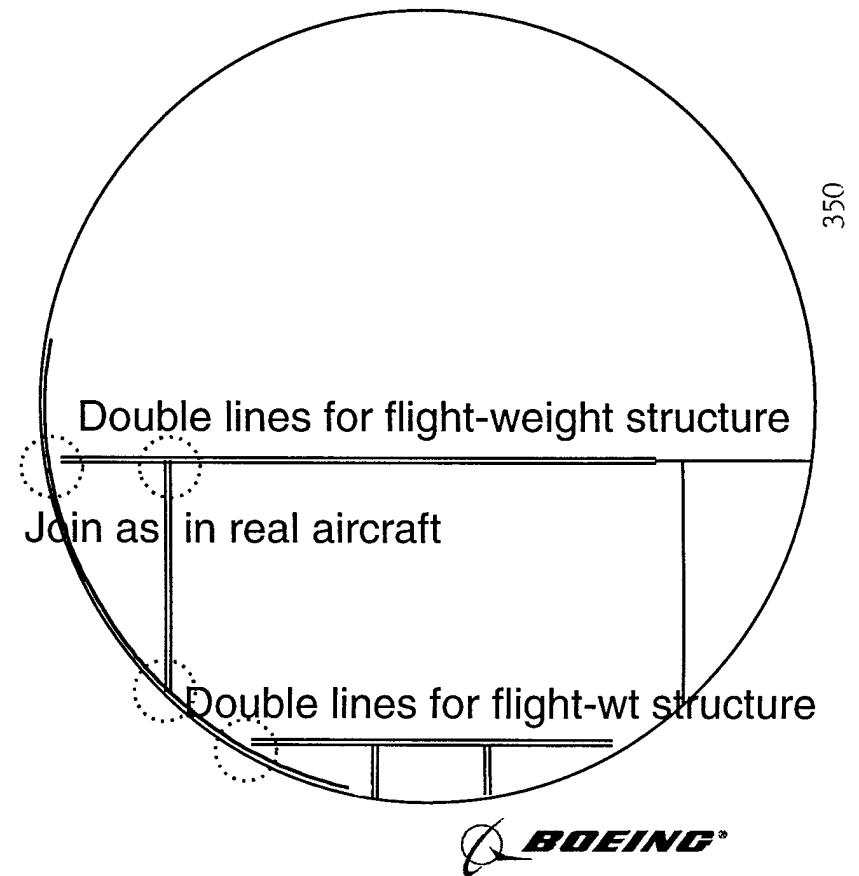
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# RBTF Basic Design Concept 1

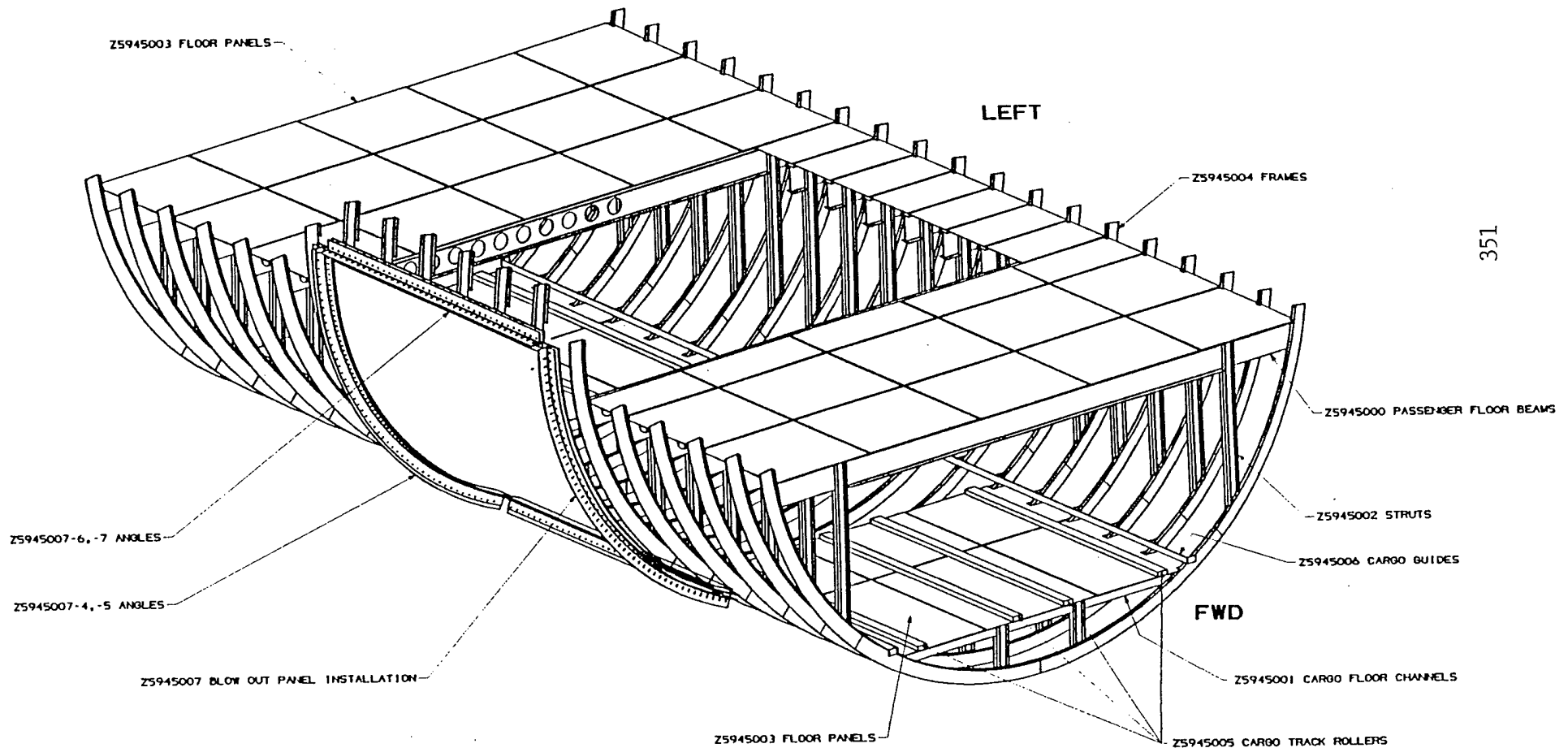
Top View



End View of Test Section



## RBTF Basic Design Concept 2



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Adding this steel design to an existing 20'-D shock-tube creates a low-cost RBTF. – The opening will be fitted with flight-weight test sections.

# Status

- Design Requirements Document Published
- PDR on 2/12/97
  - All parties satisfied with design
  - Decision to initiate fabrication prior to CDR
- Design Drawing and Analysis of Fixture Design Completed
  - Aircraft Section Patch to RBTF with M/S ~ 3
- CDR on 5/9/97
- Anticipate completion of fixture fabrication in October 97

# Summary

- Developing data and tools to properly assess threats and aircraft response to threats.